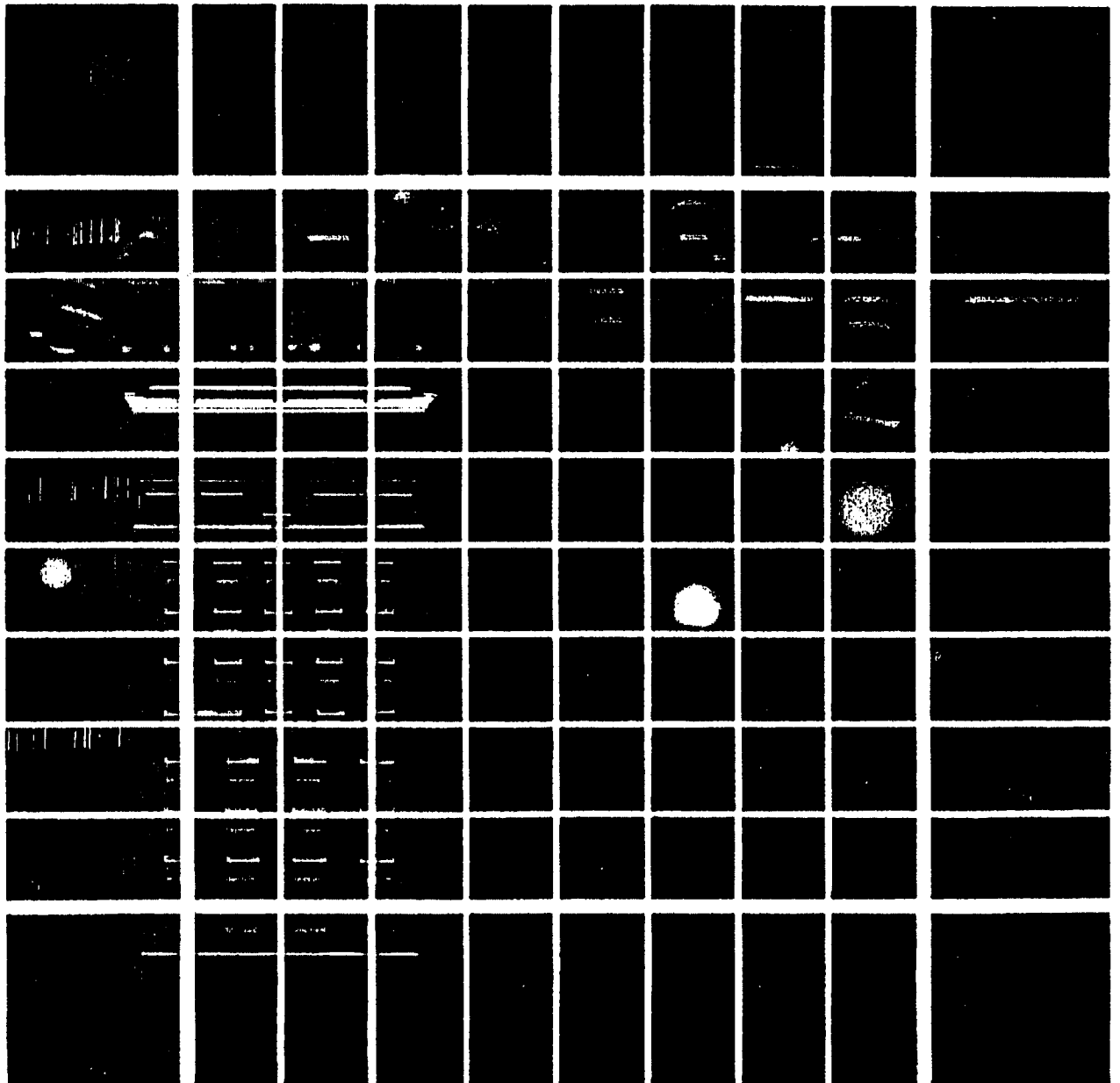


Includes barcode for easy software entry.

HEWLETT-PACKARD

HP-41

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Civil Engineering



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INTRODUCTION

This HP-41C Solutions book was written to help you get the most from your calculator. The programs were chosen to provide useful calculations for many of the common problems encountered.

They will provide you with immediate capabilities in your everyday calculations and you will find them useful as guides to programming techniques for writing your own customized software. The comments on each program listing describe the approach used to reach the solution and help you follow the programmer's logic as you become an expert on your HP calculator.

KEYING A PROGRAM INTO THE HP-41C

There are several things that you should keep in mind while you are keying in programs from the program listings provided in this book. The output from the HP 82143A printer provides a convenient way of listing and an easily understood method of keying in programs without showing every keystroke. This type of output is what appears in this handbook. Once you understand the procedure for keying programs in from the printed listings, you will find this method simple and fast. Here is the procedure:

1. At the end of each program listing is a listing of status information required to properly execute that program. Included is the SIZE allocation required. Before you begin keying in the program, press **XEQ** **ALPHA** **SIZE** **ALPHA** and specify the allocation (three digits; e.g., 10 should be specified as 010).

Also included in the status information is the display format and status of flags important to the program. To ensure proper execution, check to see that the display status of the HP-41C is set as specified and check to see that all applicable flags are set or clear as specified.

2. Set the HP-41C to PRGM mode (press the **PRGM** key) and press **■** **GTO** **■** **■** to prepare the calculator for the new program.
3. Begin keying in the program. Following is a list of hints that will help you when you key in your programs from the program listings in this handbook.
 - a. When you see " (quote marks) around a character or group of characters in the program listing, those characters are ALPHA. To key them in, simply press **ALPHA**, key in the characters, then press **ALPHA** again. So "SAMPLE" would be keyed in as **ALPHA** "SAMPLE" **ALPHA**.
 - b. The diamond in front of each LBL instruction is only a visual aid to help you locate labels in the program listings. When you key in a program, ignore the diamond.
 - c. The printer indication of divide sign is /. When you see / in the program listing, press **÷**.
 - d. The printer indication of the multiply sign is *. When you see * in the program listing, press **×**.
 - e. The † character in the program listing is an indication of the **APPEND** function. When you see †, press **■** **APPEND** in ALPHA mode (press **■** and the K key).
 - f. All operations requiring register addresses accept those addresses in these forms:
 - nn (a two-digit number)
 - IND nn (INDIRECT: **■**, followed by a two-digit number)
 - X, Y, Z, T, or L (a STACK address: **■** followed by X, Y, Z, T, or L)
 - IND X, Y, Z, T or L (INDIRECT stack: **■** **■** followed by X, Y, Z, T, or L)

Indirect addresses are specified by pressing **■** and then the indirect address. Stack addresses are specified by pressing **■** followed by X, Y, Z, T, or L. Indirect stack addresses are specified by pressing **■** **■** and X, Y, Z, T, or L.

Printer Listing	Keystrokes	Display
01♦LBL "SAMPLE"	■ LBL ALPHA SAMPLE ALPHA	01 LBL ^T SAMPLE
02 "THIS IS A"	ALPHA THIS IS A ALPHA	02 ^T THIS IS A
03 "†SAMPLE"	ALPHA ■ APPEND SAMPLE	03 ^T † SAMPLE
04 AVIEW	■ AVIEW ALPHA	04 AVIEW
05 6	6	05 6
06 ENTER†	ENTER +	06 ENTER /
07 -2	2 CHS	07 -2
08 /	÷	08 /
09 ABS	XEQ ALPHA ABS ALPHA	09 ABS
10 STO IND L	STO ■ L	10 STO IND L
11 "R3="	ALPHA R3= ■ ARCL 03	11 ^T R3=
12 ARCL 03	■ AVIEW	12 ARCL 03
13 AVIEW	ALPHA	13 AVIEW
14 RTN	■ RTN	14 RTN

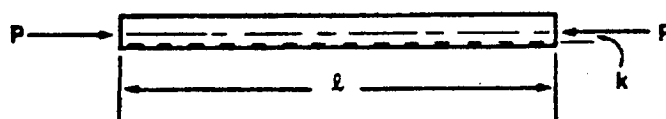
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	Calculate deflection, slope, moment, and shear.	
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	Calculate deflection, slope, moment, and shear.	
10.	BOLT TORQUE	82
	Calculate the torque needed to supply a specified load.	

STEEL COLUMN FORMULA

This program computes the allowable load and the maximum load for structural steel columns using the American Institute of Steel Construction formula (1961). The column ends must be welded, riveted, or otherwise constrained against deflection and rotation.

Equations:



$$P_{\text{allow}} = A \sigma_y [1 - (\ell/k)^2 / 2 C^2] / m \quad \text{for } \ell/k < C$$

$$P_{\text{allow}} = A(1.0273 \times 10^{12} \text{ N/m}^2) / (\ell/k)^2 \quad \text{for } C < \ell/k \leq 200$$

$$C^2 = 2 \pi^2 E / \sigma_y$$

$$m = 5/3 \times 3(\ell/k)/8C - [(\ell/k)/2C]^3$$

$$P_{\text{max}} = P_{\text{allow}} m$$

Definitions:

P_{allow} is the allowable load;

P_{max} is the maximum load the column could carry;

A is the area of the section;

ℓ is the length of the column;

k is the minimum radius of gyration of the column cross section;

I is the minimum moment of inertia of the cross section;

σ_y is the yield point of the steel.

E is the modulus of elasticity of steel.

Reference:

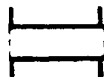
Roark, Raymond J.; Young, Warren C.; Formulas for Stress and Strain, McGraw-Hill, 1975.

Remarks:

Columns must be nominally straight, homogeneous, and of uniform cross section.

Example 1:

Two steel channels are lased together to form the cross section below:



Calculate the allowable and maximum loads using the following specifications:

$$k = 81.0 \times 10^{-3} \text{ m} \quad A = 9.46 \times 10^{-3} \text{ m}^2 \quad \sigma_y = 248 \times 10^6 \text{ N/m}^2 \quad l = 7.5 \text{ m}$$

Keystrokes:

[XEQ] [ALPHA] SIZE [ALPHA] 010

[XEQ] [ALPHA] METRIC [ALPHA]

248 [EEX] 6 [R/S]

9.46 [EEX] 3 [CHS] [R/S]

7.5 [R/S]

81 [EEX] 3 [CHS] [R/S]

[R/S]

Display:

YPS?

A?

L?

K?

Pa=918.2E3

PMAX=1.736E6

Program Listings

01♦LBL "MET RIC"	metric set-up	50 STO 08	
02 "YPS?"		51 RCL 03	
03 PROMPT		52 X>Y?	
04 STO 09		53 GTO 01	
05 207 E9		54 X<>Y	
06 X<>Y		55 200	
07 /		56 X<>Y	
08 PI		57 X>Y?	
09 X↑2		58 GTO 05	
10 *		59 SF 01	
11 ENTER↑		60♦LBL 01	
12 +		61 3	
13 SQRT		62 RCL 08	
14 STO 03		63 RCL 03	
15 10273 E8		64 /	
16 STO 07		65 X↑2	
17 GTO 00		66 STO 08	
18♦LBL "ENG LISH"	----- English set-up	67 LASTX	
19 "YPS?"		68 RDN	
20 PROMPT		69 -	
21 STO 09		70 R↑	
22 30 E6		71 *	
23 X<>Y		72 8	
24 /		73 /	
25 PI		74 5	
26 X↑2		75 ENTER↑	
27 *		76 3	
28 ENTER↑		77 /	
29 +		78 +	
30 SQRT		79 STO 06	
31 STO 03		80 FS? 01	
32 149 E6		81 GTO 00	
33 STO 07		82 RCL 09	
34♦LBL 00	----- Data prompting	83 X<>Y	
35 ENG 3		84 /	
36 "A?"		85 1	
37 PROMPT		86 RCL 08	
38 STO 01		87 2	
39 "L?"		88 /	
40 PROMPT		89 -	
41 STO 04		90 *	
42 "K?"		91 GTO 07	
43 PROMPT		92♦LBL 00	
44 STO 05		93 RCL 07	
45♦LBL A	----- Calculate	94 RCL 04	
46 CF 01	P allow	95 RCL 05	
47 RCL 04		96 /	
48 RCL 05		97 X↑2	
49 /		98 /	
		99♦LBL 07	
		100 RCL 01	

[illegible]

STEEL COLUMN FORMULA

PROGRAM REGISTERS NEEDED: 27

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ROW 1 (1 - 2)



ROW 2 (2 - 9)



ROW 3 (10 - 16)



ROW 4 (17 - 18)



ROW 5 (19 - 24)



ROW 6 (25 - 33)



ROW 7 (34 - 41)



ROW 8 (42 - 50)



ROW 9 (51 - 59)



ROW 10 (59 - 71)



ROW 11 (72 - 82)



ROW 12 (83 - 94)



ROW 13 (95 - 103)



ROW 14 (103 - 109)



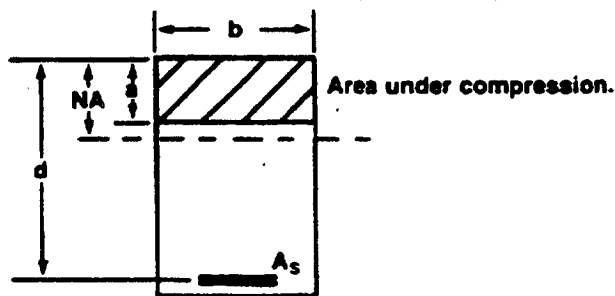
ROW 15 (110 - 112)



REINFORCED CONCRETE BEAMS

This program can be used in the design and analysis of rectangular reinforced concrete beams in accordance with the strength design method of the American Concrete Institute Code (ACI 318-71). The program solves interchangeably between the following six variables:

- A_s - The area of nonprestressed tension reinforcement (in^2 or cm^2);
- b - The width of the member (in or cm);
- M - The maximum internal bending moment (in-lb or kg-cm);
- d - The depth to the centroid of the reinforcing steel (in or cm);
- f_c - The compressive strength of the concrete (psi or kg/cm^2);
- f_y - The yield strength of the steel (psi or kg/cm^2).



During calculation of the parameters listed above, the calculator checks to be sure that enough reinforcement has been specified to meet the minimum allowable value:

$$\frac{A_s}{bd} > \frac{200}{f_y}$$

If this condition is not met the display will flash 10.50 which signifies that the design does not meet section 10.5 of the ACI code. Stop the flashing by pressing [R/S]. Press [R+] to see the current value of A_s . Press [R+] again to see the minimum allowable value of A_s .

The program also checks for too much steel. Code section 10.32 specifies the maximum steel area as:

$$\frac{A_{smax}}{bd} = (0.6375) \beta_1 \frac{f_c}{f_y} \frac{87000}{87000 + f_y}$$

where

$$\beta_1 = \begin{cases} 0.85 & \text{for } f_c \leq 4000 \\ 0.85 - (f_c - 4000)/20000 & \text{for } f_c > 4000 \end{cases}$$

If too much steel has been specified, the calculator flashes 10.32. Stop the flashing by pressing [R/S], then press [R+] to see the current steel area. Press [R+] again to see the maximum allowable tension steel area.

If the program halts displaying "DATA ERROR", the input values are mathematically impossible to satisfy. This may be due to an entry error or the configuration may be mathematically undefined. If this is the case, increase the beam size and/or decrease the moment.

Optionally, the depth of the compression zone (a) may be calculated and the depth of the neutral axis (NA) may be calculated. The depth of the neutral axis is important since T-beams may be modeled as rectangular beams if the slab or flange equals or exceeds the depth of the neutral axis.

Equations:

$$M = d \phi A_s f_y - (0.59 \phi A_s^2 f_y^2) / (b f_c)$$

$$\phi = \text{factor of safety} = 0.9$$

Reference:

ACI Standard Building Code Requirements for Reinforced Concrete (ACI 318-71), American Concrete Institute, May 1976 printing.

Remarks:

This program is intended as an aid to computation and cannot replace an understanding of ACI 318-71.

This program does not check for deflection or shear stress modes of failure. Refer to ACI 318-71 for specifics on deflection and shear stress.

Example 1:

For the specifications below, calculate the amount of reinforcing steel required.

$$M = 1.2 \times 10^6 \text{ in-lb} \quad b = 18 \text{ in} \quad d = 26 \text{ in} \quad f_c = 3500 \text{ psi}$$

$$f_y = 50000 \text{ psi}$$

Keystrokes:

Display:

[USER]	(set USER mode)
[XEQ] [ALPHA] SIZE [ALPHA] 011	
[XEQ] [ALPHA] ENG [ALPHA]	AS?
1 [R/S]	b?
18 [R/S]	M?
1.2 [EEX] 6 [R/S]	d?
26 [R/S]	FC?
3500 [R/S]	FY?
50000 [R/S]	
[A]	10.50 00

(Flashing display indicates that calculated steel area is too small to meet ACI minimum as specified in ACI 10.5. Press [R/S] to halt the flashing display. Press [R+] to see the calculated value, then press [R+] again to see the minimum value, then use the minimum value to recalculate M.)

[R/S] [R+]	1.045 00
[R+]	1.872 00
[STO] 01	
1 [STO] 03	
[C]	M=2.116E6

User Instructions

				SIZE: 011
STEP	INSTRUCTIONS	INPUT	FUNCTION	DISPLAY
1	Load program (USER mode)		[USER]	
2	Begin execution			
	English unit		[XEQ] ENG	AS?
	Metric		[XEQ] MET	AS?
	and input values	A_s	[R/S]	b?
	(input 1 for unknown value)	b	[R/S]	M?
		M	[R/S]	d?
		d	[R/S]	FC?
		f_c	[R/S]	F?
		f_y	[R/S]	
3	Calculate unknown			
	A_s		[A]	AS=
	b		[B]	b=
	M		[C]	M=
	d		[D]	d=
	f_c		[E]	FC=
	f_y		[F]	FY=
4	To change a value, store the new value and place 1.0 in the register of the unknown.			
	A_s	A_s	[STO] 01	
	b	b	[STO] 02	
	M	M	[STO] 03	
	d	d	[STO] 04	
	f_c	f_c	[STO] 05	
	f_y	f_y	[STO] 06	
		1	[STO] (unknown)	
6	To calculate depth of compressive stress			

[illegible]

Program Listings

01♦LBL "ENG	Initialization	50 GTO 01	-----
02 CF 00		51♦LBL F	f
03 GTO 00		52 XEQ 03	y
04♦LBL "MET		53 "FY="	
05 SF 00		54 6	
06♦LBL 00		55 STO 10	
07 ENG 3		56♦LBL 01	
08 "AS?"		57 RDN	
09 PROMPT		58 CHS	
10 STO 01		59 ENTER↑	
11 "b?"		60 RDN	
12 PROMPT		61 X<>Y	
13 STO 02		62 R↑	
14 "M?"		63 X↑2	
15 PROMPT		64 X<>Y	
16 STO 03		65 STO 00	
17 "d?"		66 4	
18 PROMPT		67 *	
19 STO 04		68 RCL 03	
20 "FC?"		69 *	
21 PROMPT		70 -	
22 STO 05		71 SQRT	
23 "FY?"		72 +	
24 PROMPT		73 RCL 00	
25 STO 06		74 /	
26 STOP		75 2	
27♦LBL 03	-----	76 /	
28 .59	Common to all	77 CHS	
29 RCL 05	solutions	78 GTO 00	-----
30 RCL 02		79♦LBL C	M
31 *		80 XEQ 03	
32 /		81 "M="	
33 RCL 01		82 3	
34 RCL 06		83 STO 10	
35 *		84 RDN	
36 *		85 X<>Y	
37 LASTX		86 -	
38 .9		87 GTO 00	-----
39 *		88♦LBL B	b
40 *		89 XEQ 03	
41 LASTX		90 "b="	
42 RCL 04		91 2	
43 *		92 STO 10	
44 RTN		93 GTO 01	
45♦LBL A	-----	94♦LBL E	
46 XEQ 03	A _s	95 XEQ 03	
47 "AS="		96 "FC="	fc
48 1		97 5	
49 STO 10		98 STO 10	
		99♦LBL 01	
		100 RDN	

Program Listings

101 RCL 93		151 .85	
102 -		152 RCL 05	
103 /		153 RCL 09	
104 GTO 00		154 -	
105 LBL D	-----	155 2 E4	
106 XEQ 03	d	156 /	
107 "d="		157 X<0?	
108 4		158 CLX	
109 STO 10		159 -	
110 RDN		160 STO 07	
111 X<>Y		161 .6375	
112 RCL 03		162 *	
113 +		163 RCL 05	
114 X<>Y		164 *	
115 /	-----	165 RCL 06	
116 LBL 00	Store constants	166 /	
117 STO IND		167 RCL 08	
10		168 RCL 06	
118 281		169 +	
119 STO 09		170 /	
120 6117		171 RCL 08	
121 STO 08		172 *	
122 14.06		173 X>Y?	
123 FS? 00		174 GTO 00	
124 GTO 00		175 RCL 02	
125 4 E3		176 RCL 04	
126 STO 09		177 *	
127 87 E3		178 *	
128 STO 08		179 XEQ 08	
129 200		180 -	
130 LBL 00	-----	181 RCL 01	
131 RCL 06	Check for mini-	182 10.32	
132 /	imum reinforcing	183 GTO 07	
133 RCL 01		184 LBL 00	-----
134 RCL 02		185 RCL IND	Display
135 RCL 04		10	
136 *		186 ARCL X	
137 /		187 RVIEW	
138 X>Y?		188 STOP	
139 GTO 00		189 LBL b	-----
140 RDN		190 SF 02	Calculate NA
141 LASTX		191 GTO 00	or a
142 *		192 LBL c	
143 XEQ 08		193 CF 02	
144 +		194 LBL 00	
145 RCL 01		195 RCL 01	
146 10.5	-----	196 1.18	
147 LBL 07	flashing display	197 *	
148 PSE		198 RCL 02	
149 GTO 07	-----	199 /	
150 LBL 00	check for too	200 RCL 06	
	much steel		

Program Listings

[illegible]

DATA REGISTERS				STATUS			
00	temporary storage	50		SIZE	011	TOT. REG.	62
	As			USER MODE			
	b			ENG	3	FIX	
	M			SCI		ON	X OFF
	d			DEG		RAD	GRAD
05	FC	55		FLAGS			
	FY			#	INIT S/C	SET INDICATES	CLEAR INDICATES
	l			00		metric	english
	87000 or 6117			02		a	NA
	4000 or 281						
10	control	60					
15		65					
20		70					
25		75					
30		80					
35		85					
				ASSIGNMENTS			
				FUNCTION	KEY	FUNCTION	KEY
40		90					
45		95					

REINFORCED CONCRETE BEAMS
PROGRAM REGISTERS NEEDED: 52

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ROW 1 (1 - 4)



ROW 2 (4 - 8)



ROW 3 (8 - 16)



ROW 4 (17 - 23)



ROW 5 (23 - 32)



ROW 6 (33 - 44)



ROW 7 (45 - 50)



ROW 8 (51 - 57)



ROW 9 (58 - 70)



ROW 10 (71 - 80)



ROW 11 (80 - 88)



ROW 12 (89 - 95)



ROW 13 (95 - 103)



ROW 14 (104 - 110)



ROW 15 (111 - 120)



ROW 16 (120 - 124)



ROW 17 (125 - 130)



ROW 18 (131 - 142)



REINFORCED CONCRETE BEAMS

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ROW 19 (143 - 149)



ROW 20 (150 - 156)



ROW 21 (159 - 167)



ROW 22 (168 - 179)



ROW 23 (179 - 185)



ROW 24 (185 - 192)



ROW 25 (193 - 201)



ROW 26 (202 - 209)



ROW 27 (209 - 218)

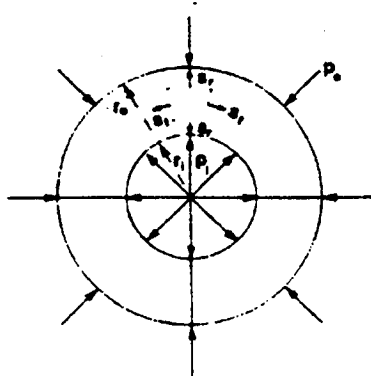


ROW 28 (218 - 222)



STRESS IN THICK-WALLED CYLINDERS

This program calculates the radial and tangential components of normal stress for thick-walled, cylindrical, pressure vessels.



Equations:

$$s_r = \frac{r_i^2 P_i - r_o^2 P_o}{r_o^2 - r_i^2} - \frac{r_i^2 r_o^2 (P_i - P_o)}{r^2 (r_o^2 - r_i^2)}$$

$$s_t = \frac{r_i^2 P_i - r_o^2 P_o}{r_o^2 - r_i^2} + \frac{r_i^2 r_o^2 (P_i - P_o)}{r^2 (r_o^2 - r_i^2)}$$

where:

- s_r is the radial component of stress;
- s_t is the tangential component of stress;
- r_i is the internal radius;
- r_o is the outer radius;
- r is the radius where calculated stresses occur;
- P_i is the internal pressure;
- P_o is the outside pressure.

Note: A negative stress indicates compression.

Reference:

J.E. Shigley
 Mechanical Engineering Design, McGraw Hill, 1963.

Example:

A cylinder has an inner radius of 1.00 inch and an outer radius of 2.00 inches. The inner pressure is 10,000 pounds per square inch and outer pressure is 150 pounds per square inch. What are the values of radial and tangential stresses for radii of 1.00, 1.25, 1.75 and 2.00 inches?

Keystrokes:

[XEQ] [ALPHA] SIZE [ALPHA] 007
 [XEQ] [ALPHA] CYL [ALPHA]
 1 [R/S]
 10000 [R/S]
 2 [R/S]
 150 [R/S]
 1 [R/S]
 [R/S]
 [R/S]
 1.25 [R/S]
 [R/S]
 [R/S]
 1.75 [R/S]
 [R/S]
 [R/S]
 2 [R/S]
 [R/S]

Display:

RI?
 PI?
 RO?
 PO?
 R?
 SR=-10.00E3
 ST=16.27E3
 R?
 SR=-5.272E3
 ST=11.54E3
 R?
 SR=-1.155E3
 ST=7.422E3
 R?
 SR=-150.0E0
 ST=6.417E3

Program Listings

01♦LBL "CYL"		51 +		
02 ENG 3		52 "ST="		
03 "RI?"		53♦LBL d		
04 PROMPT		54 ARCL X		
05 X↑2		55 AVIEW		
06 STO 00		56 RTN		
07 "PI?"		57 GTO "R"		
08 PROMPT		58 .END.		
09 STO 01				
10 *				
11 "RO?"				
12 PROMPT				
13 X↑2				
14 STO 02				
15 "PO?"				
16 PROMPT				
17 STO 03				
18 *				
19 -				
20 STO 06				
21 RCL 02				
22 RCL 00				
23 -				
24 STO 04				
25 RCL 00				
26 RCL 02				
27 *				
28 RCL 01				
29 RCL 03				
30 -				
31 *				
32 STO 05				
33♦LBL "R"				
34 RCL 06				
35 RCL 04				
36 /				
37 RCL 05				
38 "R?"				
39 PROMPT				
40 X↑2				
41 /				
42 RCL 04				
43 /				
44 -				
45 "SR="				
46 XEQ d				
47 STOP				
48 LASTX				
49 2				
50 *				

Display

$$r_1^2 P_i - r_o^2 P_o$$

$$r_1^2 r_o^2 (P_i - P_o)$$

SR

ST

70

80

90

00

[illegible]

STRESS IN THICK-WALLED
CYLINDERS
PROGRAM REGISTERS NEEDED: 14

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ROW 1 (1 - 3)



ROW 2 (4 - 11)



ROW 3 (11 - 20)



ROW 4 (21 - 33)



ROW 5 (33 - 40)



ROW 6 (41 - 48)



ROW 7 (49 - 56)



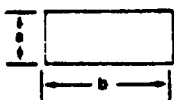
ROW 8 (57 - 58)



PROPERTIES OF SPECIAL SECTIONS

For rectangles, triangles, ellipses, circles, and concentric circles, this program performs an interchangeable solution between the section dimensions and the principle moment of inertia about the x axis. The section area and the principle moment of inertia about the y axis may also be calculated.

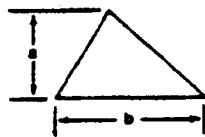
Sections and Equations:



$$I_x = a^3 b / 12$$

$$I_y = a b^3 / 12$$

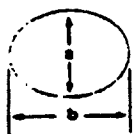
$$A = ab$$



$$I_x = a^3 b / 36$$

$$I_y = a b^3 / 36$$

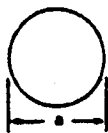
$$A = ab/2$$



$$I_x = \pi a^3 b / 64$$

$$I_y = \pi a b^3 / 64$$

$$A = \pi ab/4$$



$$I_x = \frac{\pi a^4}{64} = I_y$$

$$A = \pi a^2 / 4$$

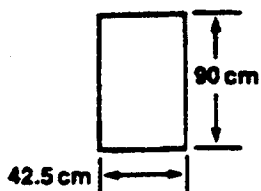


$$I_x = \frac{\pi (a^4 - b^4)}{64} = I_y$$

$$A = \frac{\pi (a^2 - b^2)}{4}$$

Example 1:

For the rectangular section below, what is the moment of inertia about the x axis? What is the moment of inertia about the y axis?



Keystrokes:

Display:

[USER]

(set USER mode)

[XEQ] [ALPHA] SIZE [ALPHA] 004

[XEQ] [ALPHA] REC [ALPHA]

a?

90 [R/S]

b?

42.5 [R/S]

IX?

[R/S]

[C]

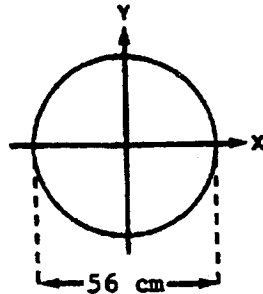
IX=2.582E6

[E]

IY=575.7E3

Example 2:

What is the moment of inertia about the x-axis for the circle below? What is the area?



Keystrokes:

Display:

[XEQ] [ALPHA] CIR [ALPHA]

a?

56 [R/S]

b?

[R/S]

[R/S]

[C]

IX=482.7E3

[D]

A=2.463E3

User Instructions

[illegible]

Program Listings

<pre> 01♦LBL "REC " 02 CF 00 03 1 04 GTO 00 05♦LBL "TRI " 06 CF 00 07 2 08 GTO 00 09♦LBL "ELL " 10 CF 00 11 3 12 GTO 00 13♦LBL "CIR " 14 SF 00 15 4 16 GTO 00 17♦LBL "CC" 18 CF 00 19 5 20♦LBL 00 21 ENG 3 22 STO 00 23 "a?" 24 PROMPT 25 STO 01 26 CLX 27 "b?" 28 PROMPT 29 STO 02 30 CLX 31 "IX?" 32 PROMPT 33 STO 03 34 STOP 35♦LBL A 36 RCL 03 37 RCL 02 38 X=0? 39 / 40 GTO IND 00 41♦LBL 01 42 12 43 GTO 00 44♦LBL 02 45 36 46 GTO 00 </pre>	<pre> Initialization </pre> <pre> Calculate a </pre>	<pre> 47♦LBL 03 48 PI 49 / 50 64 51♦LBL 00 52 * 53 3 54 1/X 55 Y↑X 56 STO 01 57 "a=" 58 GTO 10 59♦LBL 04 60♦LBL 05 61 RCL 03 62 64 63 * 64 PI 65 / 66 RCL 02 67 X↑2 68 X↑2 69 + 70 SQRT 71 SQRT 72 STO 01 73 "a=" 74 GTO 10 75♦LBL B 76 GTO IND 00 77♦LBL 01 78♦LBL 02 79♦LBL 03 80 RCL 03 81 12 82 * 83 RCL 01 84 X↑2 85 LASTX 86 * 87 / 88 XEQ IND 00 89 STO 02 90 "b=" 91 GTO 10 92♦LBL 02 93 3 94 * 95♦LBL 01 </pre>	<pre> </pre> <pre> Calculate b </pre>
--	--	--	---

Program Listings

96 RTN		146 *	
97♦LBL 03		147 36	
98 3		148 /	
99 /		149 RTN	
100 16		150♦LBL 03	
101 *		151 *	
102 PI		152 *	
103 /		153 PI	
104 "b="		154 *	
105 GTO 10		155 64	
106♦LBL 05		156 /	
107 RCL 01		157 RTN	
108 X↑2		158♦LBL 04	
109 X↑2		159♦LBL 05	
110 RCL 03		160 X<>Y	
111 64		161 X↑2	
112 *		162 *	
113 PI		163 X<>Y	
114 /		164 X↑2	
115 -		165 X↑2	
116 SQRT		166 -	
117 SQRT		167 ABS	
118 STO 02		168 PI	
119 "b="		169 *	
120 GTO 10		170 64	
121♦LBL C	-----	171 /	
122 RCL 02	Calculate I _x	172 RTN	
123 RCL 01		173♦LBL D	-----
124 XEQ 09		174 RCL 01	Calculate Area
125 STO 03		175 RCL 02	
126 "IX="		176 *	
127 GTO 10		177 GTO IND	
128♦LBL E	-----	00	
129 RCL 01	Calculate I _y	178♦LBL 02	
130 RCL 02		179 2	
131 XEQ 09		180 /	
132 "IY="		181♦LBL 01	
133 GTO 10		182 GTO 00	
134♦LBL 09	-----	183♦LBL 03	
135 ENTER↑	I _x or I _y	184 PI	
136 X↑2		185 *	
137 GTO IND		186 4	
00		187 /	
138♦LBL 01		188 GTO 00	
139 *		189♦LBL 04	
140 *		190♦LBL 05	
141 12		191 RCL 01	
142 /		192 X↑2	
143 RTN		193 RCL 02	
144♦LBL 02		194 X↑2	
145 *		195 -	
		196 PI	

197 *		51	
198 4			
199 /			
200 LBL 00			
201 "A="			
202 LBL 10			
203 ARCL X			
204 AVIEW			
205 STOP			
206 .END.		60	
20		70	
30		80	
40		90	
50		00	

REGISTERS, STATUS, FLAGS, ASSIGNMENTS

DATA REGISTERS			STATUS				
00	section code	50	SIZE	004	TOT. REG.	48	USER MODE
	a		ENG	3	FIX	SCI	ON ☒ OFF
	b		DEG		RAD	GRAD	
	Ix						
05		55	FLAGS				
			#	INIT S/C	SET INDICATES	CLEAR INDICATES	
					NONE		
10		60					
15		65					
20		70					
25		75					
30		80					
35		85					
			ASSIGNMENTS				
			FUNCTION		KEY	FUNCTION	
40		90					
45		95					

PROPERTIES OF SPECIAL SECTIONS
PROGRAM REGISTERS NEEDED: 45

HEWLETT PACKARD
SOLUTION BOOK:
CIVIL ENGINEERING

ROW 1 (1 - 5)



ROW 2 (5 - 9)



ROW 3 (9 - 13)



ROW 4 (13 - 17)



ROW 5 (17 - 24)



ROW 6 (25 - 32)



ROW 7 (33 - 42)



ROW 8 (43 - 51)



ROW 9 (52 - 61)



ROW 10 (62 - 73)



ROW 11 (73 - 81)



ROW 12 (81 - 90)



ROW 13 (91 - 101)



ROW 14 (102 - 111)



ROW 15 (111 - 120)



ROW 16 (121 - 127)



ROW 17 (127 - 133)



ROW 18 (133 - 143)



PROPERTIES OF SPECIAL SECTIONS

HEWLETT PACKARD
SOLUTION BOOK:
CIVIL ENGINEERING

ROW 19 (144 - 155)



ROW 20 (155 - 167)



ROW 21 (168 - 177)



ROW 22 (178 - 188)



ROW 23 (189 - 201)



ROW 24 (201 - 208)

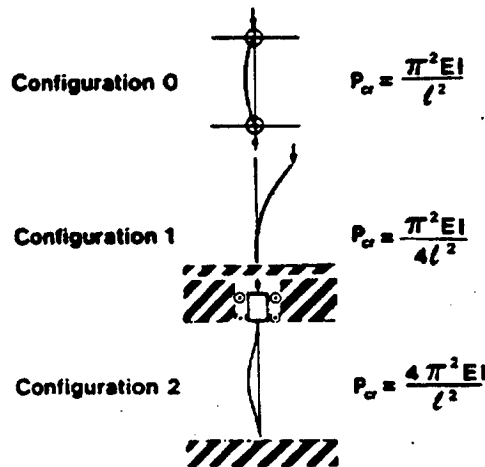


COMPRESSIVE BUCKLING

This program performs an interchangeable solution for the four properties of slender compression members or columns: P_{cr} , the critical buckling load; E , the modulus of elasticity; I , the minimum moment of inertia; and ℓ , the length of the member.

Equations:

Three configurations are possible, identified by the number of fixed ends on the member: 0, both ends hinged; 1, one end free and one fixed; 2, both ends fixed.



Remarks:

Uncertainties such as the amount of restraint at the ends, eccentricity of the load, initial warp, nonhomogeneity of the material and deflection caused by lateral loads, can cause very significant changes in the behavior of a compressive member.

Example 1:

If an 8 inch steel ($E = 30 \times 10^6$ psi) piston rod (a piston rod has zero fixed ends) must withstand a load of 15000 pounds without buckling, what moment of inertia must it have?

Keystrokes:

[XEQ] [ALPHA] SIZE [ALPHA] 007

[XEQ] [ALPHA] COMPRES [ALPHA]

0 [R/S]

15000 [R/S]

30 [EEX] 6 [R/S]

[R/S]

8 [R/S]

Display:

GEOMETRY?

P

E

I

L

I=3.242E-3

Program Listings

01♦LBL "COM
PRES"

02 ENG 3
03 SF 01
04 CF 02+
05 "GEOMETR
Y?"

06 PROMPT
07 STO 00
08 "P"
09 PROMPT
10 X=0?

11 XEQ b
12 STO 01
13 CLX
14 "E"

15 PROMPT
16 X=0?

17 XEQ a
18 STO 02
19 CLX

20 "I"
21 PROMPT
22 X=0?

23 XEQ a
24 STO 03
25 CLX

26 "L"
27 PROMPT
28 X=0?

29 XEQ b
30 STO 04
31 1

32 -
33 X=0?

34 SF 02
35 CF 00
36 "="

37 ASTO 05
38 GTO IND

00
39♦LBL 01

40 SF 00
41 XEQ 00
42 4

43 /
44 GTO d
45♦LBL 02

46 SF 00
47 XEQ 00
48 4

49 *

Initialization

one fixed end

2 fixed ends

50 GTO d

51♦LBL 00

52 PI

53 X↑2

54 RCL 02

55 *

56 RCL 03

57 *

58 RCL 01

59 /

60 RCL 04

61 X↑2

62 /

63 FS? 00

64 RTN

65♦LBL d

66 FS? 01

67 1/X

68 FS? 02

69 SQRT

70 CLA

71 ARCL 06

72 ARCL 05

73 ARCL X

74 AVIEW

75 STOP

76♦LBL b

77 CF 01

78♦LBL a

79 ASTO 06

80 1

81 RTN

82 .END.

No fixed ends

Display

unknown

90

00

[illegible]

COMPRESSIVE BUCKLING

HEWLETT PACKARD
SOLUTION BOOK:
CIVIL ENGINEERING

PROGRAM REGISTERS NEEDED: 21

ROW 1 (1 - 2)



ROW 2 (3 - 5)



ROW 3 (5 - 14)



ROW 4 (14 - 23)



ROW 5 (23 - 31)



ROW 6 (32 - 39)



ROW 7 (40 - 46)



ROW 8 (47 - 55)



ROW 9 (56 - 66)



ROW 10 (66 - 74)



ROW 11 (75 - 82)



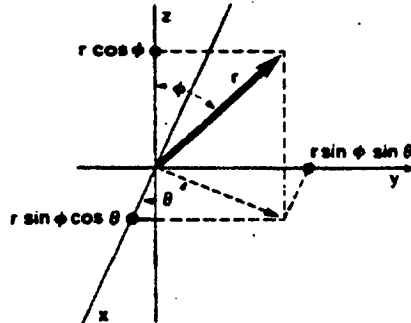
ROW 12 (82 - 82)



VECTOR OPERATIONS

This program performs the basic vector operations of addition, cross product, and dot or scalar product. It also allows conversion between spherical and cartesian coordinates and can find the angle between two vectors.

Equations:



Coordinate conversions:

$$x = r \sin \phi \cos \theta$$

$$y = r \sin \phi \sin \theta$$

$$z = r \cos \phi$$

$$r = \sqrt{x^2 + y^2 + z^2}$$

$$\theta = \tan^{-1} (y/x)$$

$$\phi = \cos^{-1} (z / \sqrt{x^2 + y^2 + z^2})$$

Vector addition:

$$\vec{v}_1 + \vec{v}_2 = (x_1 + x_2) \vec{i} + (y_1 + y_2) \vec{j} + (z_1 + z_2) \vec{k}$$

Cross product:

$$\vec{v}_1 \times \vec{v}_2 = (y_1 z_2 - z_1 y_2) \vec{i} + (z_1 x_2 - x_1 z_2) \vec{j} + (x_1 y_2 - y_1 x_2) \vec{k}$$

Dot or scalar product:

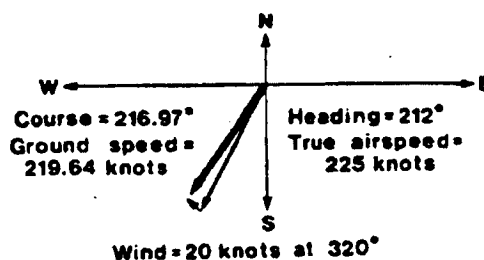
$$\vec{v}_1 \cdot \vec{v}_2 = x_1 x_2 + y_1 y_2 + z_1 z_2$$

Angle between vectors:

$$\lambda = \cos^{-1} \left(\frac{V_1 \cdot V_2}{|V_1| |V_2|} \right)$$

Example:

An aircraft flies a heading of 212 degrees at 225 knots. The wind is reported at 20 knots and 140 degrees (which translates to 20 knots and 320 degrees since winds are reported opposite to the direction they blow). What is the course of the aircraft? What is the ground speed?



Keystrokes:

[XEQ] [ALPHA] SIZE [ALPHA] 008

[XEQ] [ALPHA] SPH [ALPHA]

225 [R/S]

90 [R/S]

212 [R/S]

20 [R/S]

90 [R/S]

320 [R/S]

[XEQ] [ALPHA] ADD [ALPHA]

[R/S]

[R/S]

Display:

R?

PHI?

THETA?

R?

PHI?

THETA?

R=219.64

PHI=90.00

THETA=216.97

User Instructions

				SIZE: 008
STEP	INSTRUCTIONS	INPUT	FUNCTION	DISPLAY
1	Load program			
2	Begin execution:			
	for rectangular coordinates		[XEQ] REC	X?
	input $x_1, y_1, z_1, x_2, y_2, z_2$	x_1	[R/S]	Y?
		y_1	[R/S]	Z?
		z_1	[R/S]	X?
		x_2	[R/S]	Y?
		y_2	[R/S]	Z?
		z_2	[R/S]	
	Go to step 3			
	for spherical coordinates		[XEQ] SPH	R?
	input $R_1, \phi_1, \theta_1, R_2, \phi_2, \theta_2$			PHI?
				THETA?
				R?
				PHI?
				THETA?
	Go to step 3			
3	Execute operation			
	addition (then go to step 4)		[XEQ] ADD	
	subtraction (then go to step 4)		[XEQ] SUB	
	dot product		[XEQ] DOT	d=
	angle between		[R/S]	$\angle$ =
	cross product (then go to step 4)		[XEQ] CROSS	
4	Answers are displayed:			X= (or) R=
			[R/S]	Y= (or) PHI=
			[R/S]	Z= (or) THETA=

Program Listings

01♦LBL "SPH	SF00 for SPHERICAL	50 STO 05	
"		51 RDN	
02 SF 00		52 STO 04	
03 GTO 00		53 RDN	
04♦LBL "REC	-----	54 STO 03	
"	CF00 for Rectangular	55 STOP	
05 CF 00	-----	56♦LBL 01	input prompting
06♦LBL 00	input	57 "R?"	
07 CLRG	initialization	58 FC? 00	
08 FIX 2		59 "X?"	
09 .1		60 PROMPT	
10 STO 06		61 STO IND	
11 XEQ 01		06	
12 XEQ 01		62 ISG 06	
13 FC? 00		63 "PHI?"	
14 STOP	-----	64 FC? 00	
15 RCL 02	spherical to	65 "Y?"	
16 COS	rectangular	66 PROMPT	
17 RCL 01	conversion	67 STO IND	
18 SIN		06	
19 RCL 00		68 ISG 06	
20 *		69 "THETA?"	
21 *		70 FC? 00	
22 LASTX		71 "Z?"	
23 RCL 02		72 PROMPT	
24 SIN		73 STO IND	
25 *		06	
26 RCL 00		74 ISG 06	
27 RCL 01		75 RTN	
28 COS		76♦LBL "ADD	-----
29 *		"	CF02 for addition
30 STO 02		77 CF 02	
31 RDN		78 GTO 01	-----
32 STO 01		79♦LBL "SUB	SF02 for subtraction
33 RDN		"	-----
34 STO 00		80 SF 02	addition or subtraction
35 RCL 05		81♦LBL 01	
36 COS		82 RCL 00	
37 RCL 04		83 RCL 03	
38 SIN		84 FS? 02	
39 RCL 03		85 CHS	
40 *		86 +	
41 *		87 STO 00	
42 LASTX		88 RCL 01	
43 RCL 05		89 RCL 04	
44 SIN		90 FS? 02	
45 *		91 CHS	
46 RCL 03		92 +	
47 RCL 04		93 STO 01	
48 COS		94 RCL 02	
49 *		95 RCL 05	

Program Listings

96 FS? 02		145 RCL 00	
97 CHS		146 RCL 04	
98 +		147 *	
99 STO 02		148 RCL 01	
100 GTO 00		149 RCL 03	
101 LBL "DOT	-----	150 *	
"	Dot Product	151 -	
102 RCL 00		152 STO 02	
103 RCL 03		153 RDN	
104 *		154 STO 01	
105 RCL 01		155 RCL 07	
106 RCL 04		156 STO 00	
107 *		157 GTO 00	
108 +		158 LBL "MAG	-----
109 RCL 02		"	Magnitude
110 RCL 05		159 CF 03	
111 *		160 RCL 00	
112 +		161 RCL 01	
113 "d="	-----	162 RCL 02	
114 XEQ 04	Angle between	163 LBL 03	
115 RCL 00	two vectors	164 X↑2	
116 RCL 01		165 X<>Y	
117 RCL 02		166 X↑2	
118 SF 03		167 +	
119 XEQ 03		168 X<>Y	
120 /		169 X↑2	
121 RCL 03		170 +	
122 RCL 04		171 SQRT	
123 RCL 05		172 FS? 03	
124 XEQ 03		173 RTN	
125 /		174 "M="	
126 ACOS		175 GTO 04	
127 "Δ="		176 LBL 00	
128 GTO 04	-----	177 CF 01	
129 LBL "CRO	Cross Product	178 FC? 00	
SS"		179 GTO 00	
130 RCL 01		180 SF 24	
131 RCL 05		181 RCL 00	
132 *		182 RCL 01	
133 RCL 04		183 RCL 02	
134 RCL 02		184 SF 03	
135 *		185 XEQ 03	
136 -		186 STO 07	
137 STO 07		187 X=0?	
138 RCL 02		188 1 E-99	
139 RCL 03		189 RCL 02	
140 *		190 X<>Y	
141 RCL 00		191 /	
142 RCL 05		192 ACOS	
143 *		193 RCL 00	
144 -		194 X=0?	

195	I E-99
196	RCL 01
197	X<>Y
198	X<0?
199	SF 01
200	/
201	ATAN
202	STO 02
203	RDN
204	STO 01
205	RCL 07
206	STO 00
207	CF 24
208	FC? 01
209	GTO 00
210	180
211	ST+ 02
212	*LBL 00
213	RCL 00
214	"R="
215	FC? 00
216	"X="
217	XEQ 04
218	RCL 01
219	"PHI="
220	FC? 00
221	"Y="
222	XEQ 04
223	RCL 02
224	"THETA="
225	FC? 00
226	"Z="
227	*LBL 04
228	ARCL X
229	AVIEW
230	STOP
231	RTN
232	.END.

VECTOR OPERATIONS

PROGRAM REGISTERS NEEDED: 56

HEWLETT PACKARD
SOLUTION BOOK:
CIVIL ENGINEERING

ROW 1 (1 - 4)



ROW 2 (4 - 9)



ROW 3 (10 - 17)



ROW 4 (18 - 30)



ROW 5 (31 - 43)



ROW 6 (44 - 56)



ROW 7 (57 - 62)



ROW 8 (63 - 67)



ROW 9 (68 - 71)



ROW 10 (71 - 76)



ROW 11 (76 - 80)



ROW 12 (80 - 90)



ROW 13 (91 - 101)



ROW 14 (101 - 108)



ROW 15 (109 - 117)



ROW 16 (118 - 125)



ROW 17 (126 - 129)



ROW 18 (129 - 140)



VECTOR OPERATIONS

HEWLETT PACKARD
SOLUTION BOOK:
CIVIL ENGINEERING

ROW 19 (141 - 153)



ROW 20 (154 - 159)



ROW 21 (159 - 171)



ROW 22 (172 - 178)



ROW 23 (179 - 186)



ROW 24 (187 - 195)



ROW 25 (195 - 203)



ROW 26 (204 - 211)



ROW 27 (211 - 217)



ROW 28 (217 - 222)



ROW 29 (222 - 226)



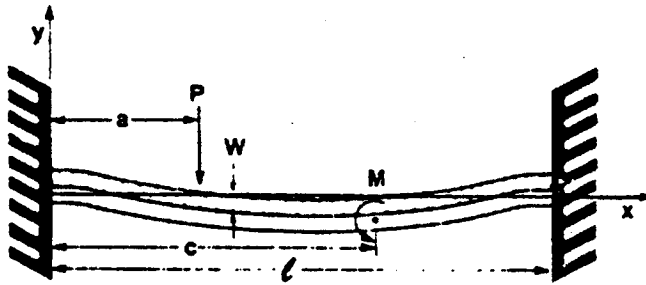
ROW 30 (226 - 232)



BEAMS FIXED AT BOTH ENDS

This program calculates deflection, slope, moment and shear at any specified point along a beam of uniform cross section, fixed at both ends. Distributed loads, point loads, applied moments or combinations of all three may be modeled. By using the principle of superposition, complicated beams with multiple point loads, and multiple applied moments can be analyzed.

Equations:



$$y = y_1 + y_2 + y_3 \quad (\text{total deflection})$$

$$y_1 = \frac{P(\ell - a)^2 x^2}{6EI\ell^3} [x(\ell + 2a) - 3a\ell]^* \quad (\text{deflection due to point load})$$

$$y_2 = \frac{Wx^2}{24EI} [x(2\ell - x) - \ell^2] \quad (\text{distributed load})$$

$$y_3 = \frac{M(\ell - c)x^2}{\ell^2 EI} \left[\frac{cx}{\ell} + \frac{\ell - 3c}{2} \right]^{**} \quad (\text{applied moment})$$

$$\theta = \theta_1 + \theta_2 + \theta_3 \quad (\text{total slope})$$

$$\theta_1 = \frac{P(\ell - a)^2 x}{2EI\ell^3} [x(\ell + 2a) - 2a\ell]^* \quad (\text{slope due to point load})$$

$$\theta_2 = \frac{Wx}{12EI} [x(3\ell - 2x) - \ell^2] \quad (\text{slope due to distributed load})$$

$$\theta_3 = \frac{M(\ell - c)x}{\ell^2 EI} \left[\frac{3cx}{\ell} + \ell - 3c \right]^{**} \quad (\text{slope due to applied moment})$$

$$M_x = M_{x1} + M_{x2} + M_{x3} \quad (\text{total moment})$$

$$M_{x1} = \frac{P(\ell - a)^2}{\ell^3} [x(\ell + 2a) - a\ell]^* \quad (\text{moment due to point load})$$

$$M_{x2} = \frac{W}{12} [6x(\ell - x) - \ell^2] \quad (\text{moment due to distributed load})$$

$$M_{x3} = \frac{M(\ell - c)}{\ell^2} \left[\frac{6cx}{\ell} + \ell - 3c \right]^{**} \quad (\text{moment due to applied moment})$$

$$V = V_1 + V_2 + V_3 \quad (\text{total shear})$$

$$V_1 = \frac{P(\ell - a)^2}{\ell^3} (\ell + 2a) \quad (\text{shear due to point load})$$

$$V_2 = \frac{-W}{2} (2x - \ell) \quad (\text{shear due to distributed load})$$

$$V_3 = \frac{-6M(\ell - c)}{\ell^3} c^{**} \quad (\text{shear due to applied moment})$$

where:

y is the deflection at a distance x from the left support;

θ is the slope (change in y per change in x) at x ;

M_x is the moment at x ;

V is the shear at x ;

I is the moment of inertia of the beam;

E is the modulus of elasticity of the beam;

l is the length of the beam;

P is a concentrated load;

W is a uniformly distributed load with dimensions of force per unit length;

M is an applied moment;

a is the distance from the left support to the point load;

c is the distance to the applied moment.

*If x is greater than a , a is replaced by $(l - a)$ and x is replaced by $(l - x)$. The signs of θ_1 and V_1 are also changed.

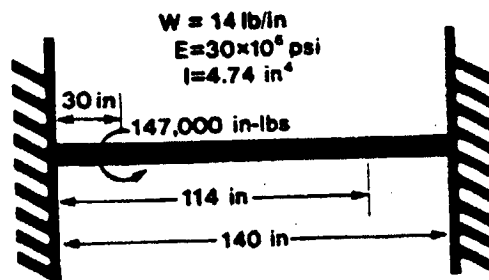
**If x is greater than c , x is replaced by $(l - x)$ and c is replaced by $(l - c)$. The signs of y_3 and M_{x3} are also changed.

Remarks:

Deflections must not significantly alter the geometry of the problem. Beams must be of constant cross section for deflection and slope equations to be valid. Stresses must be in the elastic region.

Example 1:

For the beam below, what are the values of deflection, slope, moment, and shear at an x of 114 inches?



Keystrokes:

Display:

[USER]

(set USER mode)

[XEQ] [ALPHA] SIZE [ALPHA] 011

[XEQ] [ALPHA] FIX [ALPHA]

L?

140 [R/S]

E?

Keystrokes:

30 [EEX] 6 [R/S]

4.74 [R/S]

0 [R/S]

0 [R/S]

14 [R/S]

147000 [R/S]

30 [R/S]

114 [R/S]

[A]

[B]

[C]

[D]

Display:

I?

a?

P?

W?

M?

c?

X?

Y=43.72E-3

 $\angle = -3.155E-3$

M=13.05E3

V=444.7E0

Program Listings

01♦LBL "FIX"		51 RCL 00	
"		52♦LBL 00	
02 ENG 3		53 SF 03	
03 "L?"	Initialization	54 SF 01	
04 PROMPT		55 -	angle or
05 STO 02		56 RCL 00	deflection
06 "E?"		57 *	
07 PROMPT		58 RCL 02	
08 "I?"		59 X↑2	
09 PROMPT		60 -	
10 *		61 *	
11 STO 09		62 24	
12 "a?"		63 /	
13 PROMPT		64 RCL 04	
14 STO 06		65 *	
15 "P?"		66 RCL 06	
16 PROMPT		67 XEQ 01	
17 STO 03		68 RCL 01	
18 "W?"		69 *	
19 PROMPT		70 RCL 08	
20 STO 04		71 RCL 02	
21 "M?"		72 *	
22 PROMPT		73 3	
23 STO 05		74 X<>Y	
24 "c?"		75 *	
25 PROMPT		76 FS? 00	
26 STO 07		77 LASTX	
27♦LBL E		78 FS? 00	
28 "X?"		79 -	
29 PROMPT		80 -	
30 STO 00		81 XEQ 06	
31 STOP	-----	82 6	
32♦LBL B	angle	83 /	
33 RCL 00		84 FS? 00	
34 SF 00		85 3	
35 ENTER↑		86 FS? 00	
36 +		87 *	
37 RCL 02		88 FS? 00	
38 3		89 XEQ 03	
39 *		90 XEQ 04	
40 RCL 00		91 RCL 07	
41 ENTER↑		92 XEQ 01	
42 +		93 FC? 00	
43 GTO 00		94 GTO 09	
44♦LBL A		95 3	
45 RCL 00	deflection	96 *	
46 CF 00		97 X<>Y	
47 X↑2		98 ENTER↑	
48 RCL 02		99 +	
49 ENTER↑		100♦LBL 09	
50 +		101 +	

Program Listings

102 XEQ 06		153 FS? 00	
103 XEQ 08		154 RDN	
104 RCL 09		155 FS? 00	
105 /		156 CLX	
106 "Y="		157 FS? 00	
107 FS? 00		158 RCL 08	
108 "Z="		159 6	
109 GTO d		160 *	
110 LBL D		161 X<>Y	
111 SF 00	shear	162 ENTER↑	
112 RCL 02		163 +	
113 RCL 00		164 +	
114 ENTER↑		165 *	
115 +		166 CF 03	
116 GTO 00		167 LBL 08	
117 LBL C		168 FC? 00	
118 CF 00		169 XEQ 03	
119 RCL 02	Moment	170 LBL 04	
120 RCL 00		171 RCL 10	
121 -		172 +	
122 RCL 00		173 FS? 03	
123 *		174 RTN	
124 RCL 02		175 "M="	
125 X↑2		176 FS? 00	
126 6		177 "V="	
127 /		178 GTO d	
128 LBL 00		179 LBL 01	
129 SF 03		180 CF 02	store a or c +
130 SF 01		181 STO 08	sum
131 -	shear or moment	182 RDN	
132 2		183 STO 10	
133 /		184 RCL 08	
134 RCL 04		185 RCL 00	
135 *		186 STO 01	
136 RCL 06		187 X<=Y?	x beyond loading
137 XEQ 01		188 GTO 00	point?
138 FS? 00		189 SF 02	
139 GTO 00		190 RCL 02	
140 RCL 01		191 RCL 08	
141 *		192 -	
142 RCL 08		193 STO 08	
143 RCL 02		194 RCL 02	
144 *		195 RCL 00	
145 -		196 -	
146 LBL 00		197 STO 01	
147 *		198 LBL 00	
148 FS? 00		199 RCL 05	$\frac{P(l-a)^2}{p^3}$ or
149 XEQ 03		200 FS? 01	$\frac{M(l-a)}{l^3}$
150 XEQ 04		201 RCL 03	
151 RCL 07		202 RCL 02	
152 XEQ 01		203 RCL 08	
		204 -	

Program Listings

205 FS? 01		51	
206 X↑2			
207 *			
208 RCL 02			
209 .3			
210 Y↑X			
211 /			
212 RCL 02			
213 RCL 08			
214 FS? 01			
215 GTO 00		60	
216 3			
217 *	$\frac{(l-3a) \cdot l}{2}$		
218 -			
219 RCL 02			
220 *			
221 2			
222 /			
223 RCL 08			
224 RCL 01		70	
225 *			
226 RTN	-----		
227 ♦ LBL 00	(1 + 2a)		
228 ENTER↑			
229 +			
230 +			
231 CF 01			
232 RTN	-----		
233 ♦ LBL 03	sign change		
234 FS? 02		80	
235 CHS			
236 RTN	-----		
237 ♦ LBL 06	common calculation subroutine		
238 *			
239 RCL 01			
240 ♦ LBL 05			
241 FC? 00			
242 X↑2			
243 *			
244 RTN	-----	90	
245 ♦ LBL d	display		
246 ARCL X			
247 RVIEW			
248 STOP			
249 .END.			
50		00	

DATA REGISTERS				STATUS				
00	x	50		SIZE	011	TOT. REG.	62	USER MODE
	x, (l-x)			ENG	3	FIX		ON X OFF
	l			DEG		RAD		GRAD
	P							
	W							
05	M	55						
	a							
	c							
	a, (l-a); c, (l-c)							
	EI							
10	SUM	60		#	INIT S/C	SET INDICATES	CLEAR INDICATES	
				00		angle, shear	deflection, moment	
				01		P	M	
				02		CHS	continue	
				03		used	used	
15		65						
20		70						
25		75						
30		80						
35		85						
40		90						
45		95						

BEAMS FIXED AT BOTH ENDS
PROGRAM REGISTERS NEEDED: 52

HEWLETT PACKARD
SOLUTION BOOK:
CIVIL ENGINEERING

ROW 1 (1 - 4)



ROW 2 (5 - 12)



ROW 3 (12 - 20)



ROW 4 (21 - 28)



ROW 5 (28 - 37)



ROW 6 (38 - 47)



ROW 7 (48 - 58)



ROW 8 (59 - 68)



ROW 9 (69 - 79)



ROW 10 (80 - 88)



ROW 11 (88 - 93)



ROW 12 (94 - 103)



ROW 13 (103 - 109)



ROW 14 (109 - 117)



ROW 15 (117 - 128)



ROW 16 (129 - 137)



ROW 17 (138 - 148)



ROW 18 (148 - 153)



BEAMS FIXED AT BOTH ENDS

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SOLUTION BOOK:
CIVIL ENGINEERING

ROW 19 (154 - 164)



ROW 20 (165 - 173)



ROW 21 (173 - 178)



ROW 22 (179 - 189)



ROW 23 (189 - 200)



ROW 24 (201 - 212)



ROW 25 (213 - 223)



ROW 26 (224 - 234)



ROW 27 (235 - 245)



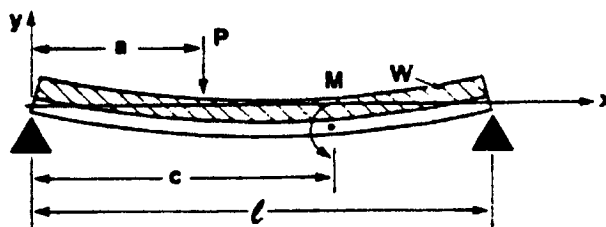
ROW 28 (246 - 249)



SIMPLY SUPPORTED BEAMS

This program calculates deflection, slope, moment and shear at any specified point along a simply supported beam of uniform cross section. Distributed loads, point loads, applied moments or combinations of all three may be modeled. By using the principle of superposition, complicated beams with multiple point loads, and multiple applied moments can be analyzed.

Equations:



$$y = y_1 + y_2 + y_3 \quad (\text{total deflection})$$

$$y_1 = \frac{P(l-a)x}{6EI} [x^2 + (l-a)^2 - l^2] * \quad (\text{deflection due to point load})$$

$$y_2 = \frac{-Wx}{24EI} [l^3 + x^2 (x - 2l)] \quad (\text{deflection due to distributed load})$$

$$y_3 = \frac{-Mx}{EI} [c - \frac{x^2}{6l} - \frac{l}{3} - \frac{c^2}{2l}] ** \quad (\text{deflection due to applied moment})$$

$$\theta = \theta_1 + \theta_2 + \theta_3 \quad (\text{total slope})$$

$$\theta_1 = \frac{P(l-a)}{6EI} [3x^2 + (l-a)^2 - l^2] * \quad (\text{slope due to point load})$$

$$\theta_2 = -\frac{W}{24EI} [l^3 + x^2 (4x - 6l)] \quad (\text{slope due to distributed load})$$

$$\theta_3 = \frac{-M}{EI} [c - \frac{x^2}{2l} - \frac{l}{3} - \frac{c^2}{2l}] ** \quad (\text{slope due to applied moment})$$

$$M_x = M_{x1} + M_{x2} + M_{x3} \quad (\text{total moment})$$

$$M_{x1} = \frac{P(l - a)x^*}{l} \quad (\text{moment due to point load})$$

$$M_{x2} = -\frac{Wx}{2} [x - l] \quad (\text{moment due to distributed load})$$

$$M_{x3} = \frac{Mx^{**}}{l} \quad (\text{moment due to applied moment})$$

$$V = V_1 + V_2 + V_3 \quad (\text{total shear})$$

$$V_1 = \frac{P(l - a)^*}{l} \quad (\text{shear due to point load})$$

$$V_2 = W \left(\frac{l}{2} - x \right) \quad (\text{shear due to distributed load})$$

$$V_3 = \frac{M}{l} \quad (\text{shear due to applied moment})$$

where:

y is the deflection at a distance x from the left support;

θ is the slope (change in y per change in x) at x ;

M_x is the moment at x ;

V is the shear at x ;

I is the moment of inertia of the beam;

E is the modulus of elasticity of the beam;

l is the length of the beam;

P is a concentrated load;

W is a uniformly distributed load with dimensions of force per unit length;

M is an applied moment;

a is the distance from the left support to the point load;

c is the distance to the applied moment.

*If x is greater than a , $(l - a)$ is replaced by $-a$ and x is replaced by $(x - l)$.

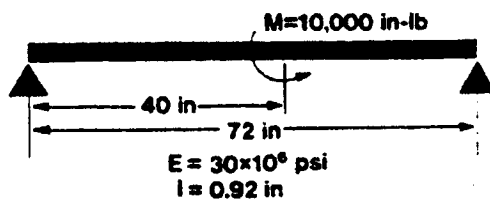
* *If x is greater than c , x is replaced by $(x - l)$ and c is replaced by $(l - c)$.

Remarks:

Deflections must not significantly alter the geometry of the problem.
Beams must be of constant cross section for deflection and slope equations to be valid. Stresses must be in the elastic region.

Example 1:

Find the deflection, slope, internal moment and shear at distances of 0, and 60 inches for the beam below. Neglect the weight of the beam.



Keystrokes:

Display:

[USER]

(set USER mode)

[XEQ] [ALPHA] SIZE [ALPHA] 011

[XEQ] [ALPHA] SIM [ALPHA]

L?

72 [R/S]

E?

30 [EEX] 6 [R/S]

I?

.92 [R/S]

a?

0 [R/S]

P?

0 [R/S]

W?

0 [R/S]

M?

10000 [R/S]

c?

40 [R/S]

X?

0 [R/S]

[A]

Y=0.000E0

[B]

 $\angle = -1.771E-3$

[C]

M=0.000E0

[D]

V=138.9E0

[E]

X?

60 [R/S]

[A]

Y=2.415E-3

[B]

 $\angle = 40.26E-6$

[C]

M=-1.667E3

[D]

V=138.9E0

User Instructions

[illegible]

Program Listings

01♦LBL "SIM		51 FS? 00	
"		52 3	
02 ENG 3	Initialization	53 FS? 00	
03 "L?"		54 *	
04 PROMPT		55 -	
05 STO 02		56 RCL 00	
06 "E?"		57 X↑2	
07 PROMPT		58 *	
08 "I?"		59 +	
09 PROMPT		60 RCL 04	
10 *		61 *	
11 STO 09		62 24	
12 "a?"		63 /	
13 PROMPT		64 RCL 00	
14 STO 06		65 X<>Y	
15 "P?"		66 *	
16 PROMPT		67 FS? 00	
17 STO 03		68 LASTX	
18 "W?"		69 CHS	
19 PROMPT		70 XEQ 01	
20 STO 04		71 RCL 01	
21 "M?"		72 X↑2	
22 PROMPT		73 FS? 00	
23 STO 05		74 3	
24 "c?"		75 FS? 00	
25 PROMPT		76 *	
26 STO 07		77 RCL 00	
27♦LBL E		78 X↑2	
28 "X?"		79 +	
29 PROMPT		80 RCL 02	
30 STO 00		81 X↑2	
31 STOP		82 -	
32♦LBL B	-----	83 *	
33 SF 00	Calculate 0 or	84 6	
34 GTO 00	Y	85 /	
35♦LBL A		86 XEQ 02	
36 CF 00		87 RCL 01	
37♦LBL 00		88 X↑2	
38 RCL 02		89 RCL 02	
39 ENTER↑		90 /	
40 *		91 6	
41 LASTX		92 /	
42 *		93 FS? 00	
43 RCL 00		94 3	
44 FS? 00		95 FS? 00	
45 4		96 *	
46 FS? 00		97 RCL 02	
47 *		98 3	
48 RCL 02		99 /	
49 2		100 +	
50 *		101 RCL 08	

Program Listings

102 XT2		153 RCL 06	
103 2		154 -	
104 /		155 STO 08	
105 RCL 02		156 RCL 00	
106 /		157 STO 01	
107 +		158 RCL 06	
108 RCL 08		159 X>Y?	
109 -		160 GTO 00	
110 *		161 RCL 06	
111 RCL 02		162 CHS	
112 *		163 STO 08	$\frac{P(l-a) x}{l}$
113 RCL 10		164 RCL 02	
114 +		165 ST- 01	
115 RCL 09		166 LBL 00	
116 /		167 RCL 03	
117 "Y="		168 RCL 08	
118 FS? 00		169 *	
119 "Z="		170 RCL 02	
120 GTO 05		171 /	
121 LBL D	----- Calculate V or M	172 FS? 00	
122 SF 00		173 RTN	
123 GTO 00		174 RCL 01	
124 LBL C		175 *	
125 RCL 00		176 RTN	
126 CF 00		177 LBL 02	----- store x and c
127 LBL 00		178 RCL 10	
128 2		179 +	
129 /		180 STO 10	
130 RCL 02		181 RCL 00	
131 FS? 00		182 STO 01	
132 2		183 RCL 07	
133 FS? 00		184 STO 08	
134 /		185 X>Y?	
135 RCL 00		186 GTO 00	
136 -		187 RCL 00	
137 RCL 04		188 RCL 02	
138 *		189 -	
139 *		190 STO 01	
140 FS? 00		191 RCL 02	
141 LASTX		192 RCL 07	
142 XEQ 01		193 -	
143 XEQ 02		194 STO 08	$\frac{m}{l}$
144 RCL 10		195 LBL 00	
145 +		196 RCL 05	
146 "M="		197 RCL 02	
147 FS? 00		198 /	
148 "V="		199 FS? 00	
149 GTO 05		200 RTN	
150 LBL 01	----- store l-a and x	201 RCL 01	
151 STO 10		202 *	
152 RCL 02		203 RTN	

[illegible]

[illegible]

SIMPLY SUPPORTED BEAMS

HEWLETT PACKARD
SOLUTIONS BOOK:
CIVIL ENGINEERING

PROGRAM REGISTERS NEEDED: 41

ROW 1 (1 : 4)



ROW 2 (5 : 12)



ROW 3 (12 : 20)



ROW 4 (21 : 28)



ROW 5 (28 : 35)



ROW 6 (36 : 46)



ROW 7 (46 : 58)



ROW 8 (57 : 67)



ROW 9 (68 : 76)



ROW 10 (77 : 87)



ROW 11 (88 : 98)



ROW 12 (99 : 111)



ROW 13 (112 : 119)



ROW 14 (120 : 128)



ROW 15 (127 : 137)



ROW 16 (138 : 145)



ROW 17 (146 : 152)



ROW 18 (153 : 164)



SIMPLY SUPPORTED BEAMS

HEWLETT PACKARD
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CIVIL ENGINEERING

ROW 19 (165 : 175)



ROW 20 (176 : 187)



ROW 21 (188 : 199)



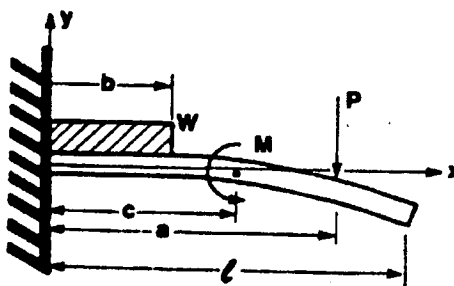
ROW 22 (200 : 206)



CANTILEVER BEAMS

This program calculates deflection, slope, moment and shear at any specified point along a rigidly fixed, cantilever beam of uniform cross section. Distributed loads, point loads, applied moments or combinations of all three may be modeled. By using the principle of superposition, complicated beams with multiple point loads, applied moments and combined distributed loads may be analyzed.

Equations:



$$y = y_1 + y_2 + y_3 \quad (\text{total deflection})$$

$$y_1 = \frac{Px_1^2}{6EI} (x_1 - 3a) - \frac{Pa^2}{2EI} (x - a) (x > a)^* \quad (\text{deflection due to point load})$$

$$y_2 = \frac{-Wx_2^2}{6EI} \left[x_2 \left(\frac{x_2}{4} - b \right) + 1.5 b^2 \right] - \frac{Wb^3}{6EI} (x - b) (x > b) \quad (\text{deflection due to distributed load})$$

$$y_3 = \frac{Mx_3^2}{2EI} + \frac{Mc}{EI} (x - c) (x > c) \quad (\text{deflection due to applied moment})$$

$$\theta = \theta_1 + \theta_2 + \theta_3 \quad (\text{total slope})$$

$$\theta_1 = \frac{PX_1}{2EI} (X_1 - 2a) \quad (\text{slope due to point load})$$

$$\theta_2 = \frac{WX_2}{EI} \left[X_2 \left(\frac{X_2}{6} - \frac{b}{2} \right) + \frac{b^2}{2} \right] \quad (\text{slope due to distributed load})$$

$$\theta_3 = \frac{MX_3}{EI} \quad (\text{slope due to applied moment})$$

$$M_x = M_{x1} + M_{x2} + M_{x3} \quad (\text{total moment})$$

$$M_{x1} = P(X_1 - a) \quad (\text{moment due to point load})$$

$$M_{x2} = -W (X_2 (X_2/2 - b) + b^2/2) \quad (\text{moment due to distributed load})$$

$$M_{x3} = M (x \leq c) \quad (\text{moment due to applied moment})$$

$$V = V_1 + V_2 + V_3 \quad (\text{total shear})$$

$$V_1 = P (x \leq a) \quad (\text{shear due to point load})$$

$$V_2 = W (b - X_2) \quad (\text{shear due to distributed load})$$

$$V_3 = 0 \quad (\text{shear due to applied moment})$$

where:

y is the deflection at a distance x from the wall;

θ is the slope (change in y per change in x) at x ;

M_x is the moment at x ;

V is the shear at x ;

I is the moment of inertia of the beam;

E is the modulus of elasticity of the beam;

l is the length of the beam;

P is a concentrated load;

W is a uniformly distributed load with dimensions of force per unit length

M is an applied moment;

a is the distance from the foundation to the point load;

b is the distance to the end of the distributed load;

c is the distance to the applied moment;

$X_1 = x$ if $x \leq a$ or a if $x > a$;

$X_2 = x$ if $x \leq b$ or b if $x > b$;

$X_3 = x$ if $x \leq c$ or c if $x > c$.

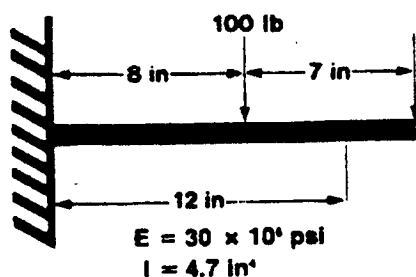
*The notation ($x > a$) is interpreted as 1.00 if x is greater than a and as 0.00 if x is less than or equal to a .

Remarks:

Deflections must not significantly alter the geometry of the problem. Beams must be of constant cross section for deflection and slope equations to be valid. Stresses must be in the elastic region.

Example 1:

What is the deflection at $x = 12$? Neglect the weight of the beam.



Keystrokes:

[USER

[XEQ] [ALPHA] SIZE [ALPHA] 011

[XEQ] [ALPHA] CANT [ALPHA]

Display:

(set USER mode)

L?

Keystrokes:

15 [R/S]

30 [EEX] 6 [R/S]

4.7 [R/S]

8 [R/S]

100 [R/S]

0 [R/S]

0 [R/S]

0 [R/S]

0 [R/S]

12 [R/S]

[A]

Display:

E?

I?

a?

P?

b?

W?

c?

M?

X?

Y=-211.8E-6

User Instructions

[illegible]

Program Listings

01♦LBL "CAN T"		51 *	
02 ENG 3		52 +	
03 "L?"	Initialization	53 RCL 03	
04 PROMPT		54 *	
05 STO 02		55 RCL 01	
06 "E?"		56 *	
07 PROMPT		57 RCL 07	
08 "I?"		58 XEQ 04	
09 PROMPT		59 RCL 07	
10 *		60 3	
11 STO 10		61 Y↑X	
12 "a?"		62 *	
13 PROMPT		63 FS? 02	
14 STO 06		64 CLX	
15 "P?"		65 STO 09	
16 PROMPT		66 RDN	
17 STO 03		67 RCL 01	
18 "b?"		68 4	
19 PROMPT		69 /	
20 STO 07		70 RCL 07	
21 "W?"		71 -	
22 PROMPT		72 RCL 01	
23 STO 04		73 *	
24 "c?"		74 RCL 07	
25 PROMPT		75 X↑2	
26 STO 08		76 1.5	
27 "M?"		77 *	
28 PROMPT		78 +	
29 STO 05		79 RCL 01	
30♦LBL E		80 X↑2	
31 "X?"		81 *	
32 PROMPT		82 RCL 09	
33 STO 00		83 +	
34 STOP		84 RCL 04	
35♦LBL A	-----	85 *	
36 RCL 06	deflection	86 -	
37 XEQ 04		87 RCL 08	
38 LASTX		88 XEQ 04	
39 *		89 6	
40 CHS		90 *	
41 3		91 RCL 01	
42 *		92 3	
43 FS? 02		93 *	
44 0		94 X<>Y	
45 RCL 01		95 FS? 02	
46 RCL 06		96 CLX	
47 3		97 +	
48 *		98 RCL 05	
49 -		99 *	
50 RCL 01		100 RCL 01	
		101 *	

Program Listings

102 +		153 GTO d	
103 6		154 LBL C	-----
104 /		155 RCL 06	Moment
105 RCL 10		156 XEQ 04	
106 /		157 RCL 01	
107 "Y="		158 RCL 06	
108 GTO d		159 -	
109 LBL B	-----	160 RCL 03	
110 RCL 06	slope	161 *	
111 XEQ 04		162 RCL 07	
112 RCL 06		163 XEQ 04	
113 2		164 CLX	
114 /		165 RCL 01	
115 RCL 01		166 2	
116 -		167 /	
117 RCL 03		168 RCL 07	
118 *		169 -	
119 RCL 01		170 RCL 01	
120 *		171 *	
121 RCL 07		172 RCL 07	
122 XEQ 04		173 X↑2	
123 RDN		174 2	
124 RCL 01		175 /	
125 6		176 +	
126 /		177 RCL 04	
127 RCL 07		178 *	
128 2		179 -	
129 /		180 RCL 08	
130 -		181 XEQ 04	
131 RCL 01		182 CLX	
132 *		183 RCL 05	
133 RCL 07		184 X<>Y	
134 X↑2		185 FS? 02	
135 2		186 +	
136 /		187 "M="	
137 +		188 GTO d	
138 RCL 04		189 LBL D	-----
139 *		190 RCL 06	shear
140 RCL 01		191 XEQ 04	
141 *		192 0	
142 -		193 FS? 02	
143 RCL 08		194 RCL 03	
144 XEQ 04		195 RCL 07	
145 RDN		196 XEQ 04	
146 RCL 05		197 CLX	
147 RCL 01		198 RCL 01	
148 *		199 RCL 07	
149 +		200 -	
150 RCL 10		201 RCL 04	
151 /		202 *	
152 "Z="		203 -	

Program Listings

204 "V="		51	
205 GTO d			
206♦LBL 04			
207 CF 02			
208 RCL 00			
209 STO 01			
210 X<>Y			
211 X<=Y?			
212 STO 01			
213 X>Y?		60	
214 SF 02			
215 -			
216 RTN			
217♦LBL d	-----		
218 ARCL X	Display		
219 AVIEW			
220 STOP			
221 .END.			
		70	
30		80	
40		90	
50		00	

REGISTERS, STATUS, FLAGS, ASSIGNMENTS

DATA REGISTERS			STATUS			
00	x	50	SIZE <u>011</u> TOT. REG. <u>54</u> USER MODE ENG <u>3</u> FIX <u> </u> SCI <u> </u> ON <u>X</u> OFF <u> </u> DEG <u> </u> RAD <u> </u> GRAD <u> </u>			
	x''(a)					
	l					
	P					
	W					
05	M	55	FLAGS # INIT S/C SET INDICATES CLEAR INDICATES			
	a		02		x > a	x < a
	b					
	c					
	temp storage					
10	FI	60				
15		65				
20		70				
25		75				
30		80				
35		85				
			ASSIGNMENTS			
			FUNCTION		KEY	
40		90				
45		95				

CANTILEVER BEAMS

PROGRAM REGISTERS NEEDED: 44

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ROW 1 (1 - 3)



ROW 2 (4 - 12)



ROW 3 (12 - 19)



ROW 4 (20 - 27)



ROW 5 (27 - 35)



ROW 6 (36 - 45)



ROW 7 (46 - 58)



ROW 8 (58 - 68)



ROW 9 (69 - 79)



ROW 10 (80 - 90)



ROW 11 (91 - 102)



ROW 12 (103 - 110)



ROW 13 (111 - 121)



ROW 14 (122 - 132)



ROW 15 (133 - 144)



ROW 16 (144 - 153)



ROW 17 (153 - 162)



ROW 18 (163 - 173)



CANTILEVER BEAMS

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ROW 19 (174 - 184)



ROW 20 (185 - 191)



ROW 21 (191 - 199)



ROW 22 (200 - 207)



ROW 23 (208 - 218)



ROW 24 (218 - 221)



BOLT TORQUE

This program may be used to calculate either the torque that will yield a specified bolt load or the load resulting from a specified torque. The maximum shear stress in the body of the screw may also be calculated.

Equations:

$$T = W \frac{D_m}{2} \left[\frac{\tan \alpha + f_t / \cos \theta}{1 - f_t \tan \alpha / \cos \theta} \right] + W f_c \frac{D_c}{2}$$

$$\tau_{\max} = \sqrt{(W / 2 A_r)^2 + (16 T_t / \pi D_r^3)^2}$$

$$T_t = T - W f_c \frac{D_c}{2}$$

where:

- T is the applied torque;
- W is the bolt load;
- D_m is the mean thread diameter;
- α is the helix angle of the thread;
- f_t is the coefficient of thread friction;
- θ is one-half of the thread angle;
- f_c is the collar coefficient of friction;
- D_c is the collar diameter;
- $\tau_{\max}$ is the maximum shear stress in the body of the screw;
- A_r is the root area;
- D_r is the diameter at the root of the thread.

Note:

The accuracy with which f_t and f_c are approximated has a significant effect on the applicability of the resulting computations.

References:

Hall, Holowenko, Laughlin Machine Design, Schaum's Outline Series, McGraw-Hill Co., 1961.

Example:

Some bolts must exert a force of 11,000 pounds each. What torque is necessary to achieve this load assuming the following specifications? What is the shear stress in the bolt?

$$\begin{array}{ll} D_m = 0.3344 \text{ in} & f_c = 0.30 \\ \alpha = 3.40^\circ & D_c = 0.8750 \\ f_t = 0.15 & D_r = 0.2983 \\ \theta = 30^\circ & \end{array}$$

Keystrokes:

[XEQ] [ALPHA] SIZE [ALPHA] 010

[XEQ] [ALPHA] BOLT [ALPHA]

3.4 [R/S]

30 [R/S]

0.15 [R/S]

0.3344 [R/S]

0.8750 [R/S]

0.3 [R/S]

11,000 [R/S]

[R/S]

[XEQ] SHEAR

0.2983 [R/S]

Display:

a?

Δ ?

FT?

DM?

DC?

FC?

W?

T?

T=1876.03

DR?

TMAX=114,335.98

[illegible]

Program Listings

01♦LBL "BOL T"	Initialization	51 RCL 01	
02 FIX 2		52 RCL 06	
03 "a?"		53 *	
04 PROMPT		54 RCL 04	
05 TAN		55 RCL 05	
06 STO 03		56 *	
07 "<?"		57 +	
08 PROMPT		58 /	
09 COS		59 STO 07	
10 STO 02		60 "W="	
11 "FT?"		61 GTO 05	
12 PROMPT		62♦LBL B	----- Calculate T
13 STO 01		63 RCL 01	
14 RCL 02		64 RCL 06	
15 /	----- Common to both W + T	65 *	
16 RCL 03		66 RCL 04	
17 +		67 RCL 05	
18 1		68 *	
19 RCL 01		69 +	
20 RCL 03		70 RCL 07	
21 *		71 *	
22 RCL 02		72 STO 08	
23 /		73 "T="	
24 -		74 GTO 05	
25 /		75♦LBL "SHE	----- Calculate maximum shear stress
26 STO 01		AR"	
27 "DM?"		76 "DR?"	
28 PROMPT		77 PROMPT	
29 2		78 2	
30 /		79 PI	
31 STO 06		80 /	
32 "DC?"		81 X<>Y	
33 PROMPT		82 /	
34 2		83 LASTX	
35 /		84 /	
36 STO 05		85 8	
37 "FC?"		86 LASTX	
38 PROMPT		87 /	
39 STO 04		88 RCL 01	
40 "W?"		89 RCL 06	
41 PROMPT		90 *	
42 STO 07		91 RCL 07	
43 CLX		92 *	
44 "T?"		93 *	
45 PROMPT		94 RCL 07	
46 X=0?		95 R-P	
47 GTO B		96 X<>Y	
48 STO 08		97 RDN	
49♦LBL A		98 *	
50 RCL 08	----- Calculate W	99 "TMAX="	-----
		100♦LBL 05	Display

Program Listings

101 ARCL X		51	
102 RVIEW			
103 STOP			
104 .END.			
10		60	
20		70	
30		80	
40		90	
50		00	

[illegible]

BOLT TORQUE

PROGRAM REGISTERS NEEDED: 23

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ROW 1 (1 - 3)



ROW 2 (4 - 11)



ROW 3 (12 - 24)



ROW 4 (25 - 32)



ROW 5 (32 - 40)



ROW 6 (40 - 48)



ROW 7 (48 - 60)



ROW 8 (60 - 89)



ROW 9 (70 - 75)



ROW 10 (75 - 81)



ROW 11 (82 - 94)



ROW 12 (95 - 101)



ROW 13 (102 - 104)



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